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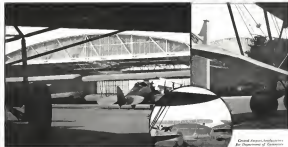
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The Olden American Aeronautical Magazine

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Editor

DAVID J. LARKE
Publishing Editor

CHARLES E. GALE

DAVID J. LARKE

JAMES E. McGRATH, Jr.
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Web with white rubber or rubber (green) 1 or 2 shade optically ground and polished meniscus lenses 21.00
Web with white rubber or rubber (green) NON-SHATTERABLE meniscus lenses 36.00
For Black Oxford Finish add \$1.50 to the above prices.



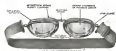
MEYROWITZ LUXOR GOGGLE U.S. Air Service Model No. 7

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Web with white optically ground and polished lenses in accordance to U. S. Army Specification No. 3114-B (1) one lens is broken, the other must be returned to be matched 19.00
Web with white hand ground and polished meniscus lenses in accordance to Navy Specification No. 341-B 22.00
Web with white hand ground and polished NON-SHATTERABLE meniscus lenses 36.00



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Web with white rubber or rubber (green) 1 or 2 shade optically ground and polished cylindrical lenses 15.50
Web with white optically ground and polished lenses in accordance to U. S. Army Specification No. 3114 17.00
Web with white optically ground and polished lenses in accordance to U. S. Army Specification No. 3114-B (1) one lens is broken, the other must be returned to be matched 18.00
Web with white hand ground and polished meniscus lenses in accordance to Navy Specification No. 341-B 21.00
Web with white hand ground and polished NON-SHATTERABLE meniscus lenses 35.00



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Web with white hand ground and polished meniscus lenses 11.00
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Web with white optically ground and polished lenses in accordance to U. S. Army Specification No. 3114 14.00
Web with white optically ground and polished lenses in accordance to U. S. Army Specification No. 3114-B (1) one lens is broken, the other must be returned to be matched 15.00
Web with white hand ground and polished meniscus lenses in accordance to Navy Specification No. 341-B 18.00
Web with white hand ground and polished NON-SHATTERABLE meniscus lenses 32.00

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AVIATION

THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

A Monthly Film Department ESTABLISHED 1910

EDWARD P. WARNER, Editor

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Shows or "Shows"?

TALK about shows and you are sure of an audience. For some months past, that subject has been an addling bit with any aeronautical gathering. Are there too many shows? Are there enough? Are they right kind? Where should they be held? What? Why? As so on, ad infinitum.

Before contributing to the wilderness of words that eddies presently around the subject, we propose to take a lesson from an individual who lived four thousand miles from Kitty Hawk and died two thousand years before the airplane left the realm of mythology. "First" said Socrates, and he said it with such persistence that posterities died at his approach, "define your terms." More colloquially, "whaddya mean, show?" What is an aircraft show, and what ought it to be? Has it always been, or should it be, and should the name ever change? If we had a clearer idea of just what it is that we are talking about, and of just what ideal we ought to have before us in all this discussion, we would be in a better position to decide intelligently how many shows are worth the industry's while, and how much money can properly be spent upon them. We have visited, from time to time, aircraft shows variously designated to: (a) Sell airplanes at retail to the public; (b) Sell airplanes at wholesale to distributors; (c) Make the population of the village "air-minded"; (d) Provide an added halcyon for a radio exhibit that was threatening to drag another financially; (e) Give engineers an opportunity of comparing the latest technical developments; (f) Support a worthy charity; (g) Live the pockets of the professional promoters; and so on, also ad infinitum. All very worthy aims, but not all of equal importance to the aircraft industry.

If our object be to compare notes on progress in design, to provide a convenient occasion for the gathering of the class, or to result a group of important trans-

port operations and super distributors, we may reasonably fix attention upon a single national show, or two at most, one east of the Rockies and the other upon the Pacific coast. To accomplish any enterprise on a truly and uniformly national scale is a matter of extraordinary difficulty in a country the size of this one, and a large proportion of the planes and engines displayed on the west coast were absent from the recent St. Louis and Detroit exhibitions.

If, on the other hand, we are trying to sell airplanes as the general public we must go where they are to be found. We must take the shows to their home towns and hold not merely three or four exhibitions but something between a score and a hundred. If the object of such local displays is to sell airplanes there should be an effort to attract the attendance of those who are, and seem likely to remain far from the financial straits of society in which planes may be bought. If what we are seeking is the production of air-mindedness, with the ultimate object of improving the unimproved airport or securing more liberal patronage for the air road all corners are welcome and any sort of a side-show thus increases the attendance is helpful.

One national show, or at most two in widely-separated parts of the country, is quite enough. The industry must not make more radical present conditions, but there need be no suppression of local exhibitions, planned for quite a different purpose. The objection to such shows is the past has been to their refusal to stay local. Every community has tried to make it clear that the show was to be the biggest on earth. It has been a poor-spirited Chamber of Commerce that could not conceive more such descriptive adjective as international or superlatively national, and thus the screws have been put on the poor aircraft industry to help make good.

The Aeronautical Chamber of Commerce ought to

back as members of working assemblies, first against any such pressure. Its first step is clearing up the situation should be the admission of the Class B function. There should be an transition stage between the view of first-rate national importance and that which is purely local and supported only by the local dealers. The first group must be extraordinarily select, more so than it has been this year. The second can vary its dimensions to meet the needs of the moment. Dealer shows, which can and should be staged very modestly and unexpensively, belong on the calendar wherever the manufacturers' local representatives want them. There should be the invitation of support, and they should get the kind of show they want to pay for, and get it when they want it. Selection must then be confined to the management's making appeal for exhibits only, and marketing representatives only with the local people.

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The Factory Learns from the Field

AVIATION, perhaps, is a greater client than any other industry, has taken advantage of the value of direct shop contacts for the designers and draftsmen. Involving as it did its intimate and expert knowledge of so many basic industries it was, but natural that aircraft building should develop the fullest degree of contact between designer and production man. The draftsman has been allowed to look from his drafting board to see an incorrect general knowledge that would provide the key to economical production. It has paid well to send theory and practice far in the present state of the industry's development good dated designs in the direct consequence of having a well-informed drafting force.

There is, however, a growing need for the extension of this drafting contact to the flying field. It is not enough for major executives to avoid the virtues of observability. That again is a fatal problem only successfully met by the transformation of many personal observations and first-hand experiments into a composite whole.

Only with amazing infrequency do aeronautical draftsmen get it as an experience at a flying field.

Major overhauls of plane and engine often done at the manufacturer's plant involve a great deal of useful information to the draftsmen but the really significant phases of everyday serviceability, only brought out at the operation field itself, are overlooked. Chief engineers, to be sure, usually have more than a passing acquaintance with actual flying operations, but theirs are the broader basic problems of design and they can hardly be expected to find time to pay a great deal of attention to the details of the tribulations of the "grease monkey."

Yet the humble service mechanic is a powerful influence. The prospective plane purchaser, perhaps none

less, immediately acquainted with aviation may occasionally seek some first-hand information at the operation field and to make the enthusiasm of a pilot adviser. To him the service mechanic may represent direct expert advice. The fate of an impending sale of a particular type of plane with which the mechanic has been struggling all according to intricate some wholly inaccessible points and trying to induce a particularly hard-to-reach tin bugles in the balance.

It is not only a question of the commencement of the mechanic. If necessary service is made too difficult of accomplishment it is always expensive and it is often likely to be neglected or overlooked. On such detail attention depends the length of life and the reliability of a plane and prospective purchasers of planes, of auto-repairs are looking rather closely into questions of "serviceability" when buying.

Securing service contacts for the draftsman is not as easy as getting him out into the shop, for there is less direct access for his presence at the field during working hours. It is, however, possible by proper consideration, to encourage him to spend some of his leisure time in contact with actual flying operations. He should be encouraged both morally and financially to seek some flying training either through the medium of an actual flying instruction course or through a flying club. If that cannot be arranged, he should be made to feel welcome at all times at the flying field, with the hope held out that in any flying demonstration or test in which these new facilities appeared for additional passengers he will be on the preferred free list.

The proportion of aeronautical draftsmen who now never even flows in a plane is as far too large for the general good of the industry. Of great import for the improvement of design from the point of view of service is the incorporation of draftsmen and plane designers with those field atmosphere. Serviceability becomes an increasingly important aspect of aircraft design. Do nearly a direct problem it should be kept before the eyes of those most intimately associated with design details—the draftsmen.

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On Racing Rules

THE ICA is a dilemma that confronts all mechanical sports motor bearing automobile racing and mechanically oriented have been common sufferers. No one has ever yet devised a satisfactory means of ruling internal combustion engines for competition which will at the same time be simple in application make bribery difficult and avoid the encouragement of freak design and fit for any conceivable application.

The airplane is set upon the same path that has been fraught with so much misapprehension and such unexpected and unfortunate results, for high speed vehicles of the

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water and of the highway. There may be no reasonable escape from that course, but at least we should realize where we are going.

For a number of years almost since the beginning of the internal car race, competing phases have been grouped in classes in terms of the displacement of their engines. That has been strictly in the automobile tradition, and the inherent result will, we predict, here be met with automobile experience. As competition in the industry becomes keener, and as racing becomes the object of more serious solicitude on the part of manufacturers we shall see engines designed, built and put through their Department of Commerce tests with emphasis in speed competition as their ultimate object and with the bearing of the existing racing rule as the fundamental motive in their design.

When racing started on the speedway at Indianapolis, the competing cars were fitted with engines not dissimilar from those of the passenger vehicles in general use at the time. With the passage of years, the racing car and its power-plant have been developed along lines entirely distinct from anything that finds its way upon the highway. A speed of 6000 r.p.m. has become commonplace. Super-charging has been carried to the limit possible without continuous deterioration. Better fuels have been refined and more doped, has been added to them and the super-charger has been stepped up again. The Indianapolis engines of 1929 were developing approximately 1.7 hp per cubic inch of piston displacement, or from seven to ten times a good commercial standard.

The same thing has happened upon the water. Except in the outboard engine field, where the governing body has waged a constant struggle against specification racing design by the enforcement of elaborate technical stipulations, engines have increased in performance and in cost as rapidly as they have diverged from anything that can be considered suitable for general use.

Perhaps the parallel is not yet apparent for the airplane, but still we foresee that a continuance of the piston displacement rating rule is going to lead within the next few years to the production of engines of small displacement, with very high compression ratios and supercharged to the limit, running at immense speed and geared down to the propeller, far too heavy and too wasteful of fuel for practical requirements, just barely able to fly through a Department of Commerce test (favoring that as approved type certificate continues to be required for all racing machines) but putting out a relatively enormous power per inch of volume.

Limitation of engine weight also has been suggested as an alternative for the displacement system and this method actually has been employed in some of the high plane contests abroad.

In aviation, the high performance engine, running supercharged and with special fuels and thereby reducing its weight and frontal area, has a larger plane than in any other field, but its development may easily run too

far in the wrong direction, and too much cost must be sacrificed on the altar of maximum power output.

Choosing prognosticated gloomily for the outcome of the present practice, we frankly admit that we do not know just what to do about it. An obvious alternative is to classify by piston displacement, but, equally obvious as an effect of that plan, propellers would then be cut down in diameter or reduced in pitch angle and the engines would be run to far beyond their rated r.p.m. and power. Racing engines, in fact, might well be deliberately given a low rating by taking their Department of Commerce test at a reduced r.p.m., in anticipation of just such an increase in power output for competitive purposes.

Limitation of fuel is a possibility, and it has been successful in certain European automobile and automobile road races in the past. Instead of limiting to 800 cubic inches of piston displacement, the limit might be to one gallon of fuel for every ten miles. A scheme theoretically sound but full of practical drawbacks. The opportunity for fraud is obvious, pointers are sure to arise, and the winner of a race can never be determined until the contents of all fuel tanks have been checked up and the penalties levied for minor excesses of consumption have been applied. As a practical matter it must be shorn out of consideration as a ruling for general use.

A rule might be made, and on the whole we think this is the most promising suggestion, that all engines used must not only have received their type approvals but also have been built to the number of at least fifty (50) examples and installed in a like number of aeroplanes. This, obvious that there is that very interesting and truly commercial engines in their developmental stage would be relegated to the experimental races. To be quite qualified none of these alternatives satisfy us at all. For the present the continuance of the flat piston displacement rule need hardly be challenged. In the future, we are very much afraid it is going to make some trouble and encourage undesirable design tendencies. At least, as potential consequences demand careful consideration by the industry, and a substitute may thus be brought to light.

Whatever rule is adopted, it should be left as nearly as possible static from year to year. In the past, the classifications of piston displacement have been governed largely by the volume of certain engine types in extensive use and, in particular, they have worked up and down from the 500 cubic inches of the C-5. For this year, as attempt has been made to serve as a logical grouping of size limitations. They should be carefully considered, modified where necessary, then allowed to remain as standard. Let the critic speak promptly, or else prepare to hold his peace. For the future, as racing should follow the rule established by the Indianapolis speedway, that no change in fundamental technical specifications affecting the design of planes and engines should become effective in less than a year, or even two years, after notice.

IMPRESSIONS OF THE *New* York Show

By LESLIE E. NEVILLE
Technical Editor of AVIATION

Exposition Held Last Week at Madison Square Garden Included Forty-Seven Planes of Which Six Were Large Transports

TO THE VISITING MEMBERS of the public the New York Aircraft Show was a rather interesting affair, but to the visiting members of the industry it was just another air show. In fact, it might be stated that the New York exhibit proved to be somewhat of a disappointment. However, it should not be taken that the failure of the show to come up to expectations was the result of poor or inadequate management. The Show Committee worked hard and, we believe, left no stone unturned to make the Show a success.

Of course the main idea of the Show was to further acquaint the public with the airplane as a means of satisfactory and economic transportation. And in that the

Show more or less filled its purpose. As one entered Madison Square Garden one was immediately confronted with a display of six great transport planes of the types which are now being used on various American airlines. The six planes were the Polder P-42, the S-AT Ford Tri-motor, the Sikorsky S-38, the Curtiss "Condor," the Consolidated "Commodore," and the Savoia-Marchetti S-55. Unfortunately, though, the size of the main arena did not permit a really good display of these great airplanes. Wings overlapped wings, rudders and elevators had to be adjusted to clear parts of other planes, and tails and fuselages were projecting out so far that visitors were forced to duck when attempting to pass by.

That condition, perhaps the worst of the whole show existed in both the main arena and the exhibition hall in the basement. In short there were far too many planes crowded into the show with the result that no plane received the advantages of display to which it was entitled. If future shows are planned for Madison Square Garden it will undoubtedly be borne in mind by the Show Committee that a total of 47 planes, six of which are big transports, is too much wing and fuselage area.

The accessory and power-plant exhibits while not grudgingly few when compared with aircraft shows of the past, were well arranged from the standpoint of decor-

ation, but their location in the building can be considerably improved upon in future.

One feature of the Show which was welcomed to be for the sole benefit of the public, but which resulted in being very much of a nuisance, was the broadcasting of announcements and speeches throughout the building. When one attempted to converse with a friend or business acquaintance he was immediately drowned out by the "blatting" of the radio amplifiers. And when one endeavored to concentrate on what was "coming over the air" the effort was to little avail. Perhaps it was the noisiness of the building or poor articulation, but at any rate it was extremely difficult, and sometimes practically



View of the transport exhibit in the arena from the eastern end showing the Polder P-42 in the foreground and the P-41 suspended above.

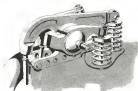
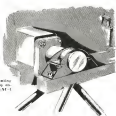


The transport exhibit in the basement entry showing the Ford tri-motor at the right and the Savoia-Marchetti S-55 directly behind it.



The main shaft of the Wright flying plane.

Detail of mounting the Wright flying machine in the S-51.



Vertical shaft of the General Air Machine Co. engine.



The transverse profile of the Washington "Sunbeam" wing, made visible above and below.



Detail sketch of the wing attachment and motor on the side of the Wright Model 5-51.

impossible to maintain the speed before the "take-off". At the present writing official attendance figures are not available. However, it is believed that total attendance did not establish any records. If that eventually proves to be the case it can be largely attributed to the weather during Show Week. After weeks of unusually raw weather for spring in New York a heat wave suddenly descended upon Manhattan and baked man, beast, bird and fish in preparation. A day or two after the arrival of the heat wave the Show opened, and for the rest of the week Madison Square Garden was little short of a Turkish bath. Perhaps that experience will lend strength of argument to those who believe that the New York show should always be held during January or February.

There is considerable difference of opinion regarding the value of the New York affair from the standpoint of sales. In the opinion of some exhibitors the show was a waste of time and money. Yet some others report that they received more good leads at the New York Show than at any other aeronautical function. It should prove interesting to judge the truth of such statements by the number of exhibitors who apply for space at the next New York Show sponsored by the Aeronautical Chamber of Commerce. However, the Chamber has been recommending an aero rendezvous for some time and any sales failure of the New York show should not be laid at its door.

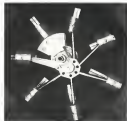
TECHNICAL progress has not been able to keep up with the rapidly recurring annual shows of the year and as a result there was little new to be exhibited.

One of the significant features bearing out former predictions was the great proportion of amphibians and flying boats. Of the 47 planes actually exhibited ten were water craft, equally divided as to amphibian or flying boat. The trend toward flying boats has also entered the glider field and was manifested in the form of the Post Office glider. This craft was one of the six amphibious machines exhibited and is designed to be towed behind a speed boat to attain flying speed. Of the 47 planes there were 23 monoplanes, an equal number of biplanes and a single sesquiplane. While the large majority were powered with radial engines, there were seven representatives of the in-line type: the Aerocraft with its opposed engine and the Curtiss Condor representing the water-cooled type.

One of the weaknesses that cannot be overlooked is the effect of the low pressure line on the leading gear design in the light and modern weight plane class. A number of the newer light planes are showcasing the shock absorber entirely where low pressure area are used, resulting in poorer landing gears and greater aerodynamic efficiency. However, some structural economy is sacrificed as it most cases the designing engineers have been faced with the necessity of increasing struc-

tural weight slightly to provide sufficient strength for landing stresses for this type of gear. The original and duplicate Packard Thorndyke engines were shown and a special showing of Packard power plants was staged at the New York showrooms of the Packard Motor Car Co.

Two or three air airplanes in the transport exhibit in the room were exhibited for the first time at the New York Show. Of these the Consolidated Commodore has been described in detail in the January 12, 1935 issue of AVIATION. The second, which is sched-

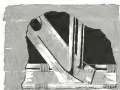


The structure of the Curtiss Condor and its assembly.

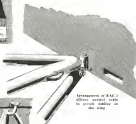
uled for detailed description in the near future, is the American built Sencos-Marchetti S-35. The S-35 is a two-hull tandem engine flying boat of all wood construction and the main deck at the show was covered with two Wright engines, the power plant being located well above the wing. The design embodies the use of a three-spar, monocoque wing and tail surfaces attached to the hull by



The Curtiss Condor is to be produced as an amphibian by the Curtiss Aeroplane and Motor Co.



A portion of the interior of the Fieseler Storch 1-40



Arrangement of EAC-1 offset control sticks to prevent stalling at the wing



The shape-changing device on the stabilizer of the new Mustang



Fuselage section showing the shape-changing wing

means of outriggers. The plane has a wing span of 28 ft. 9 in., an overall length of 34 ft. 11 in. and a wing area of 997.7 sq. ft. The passenger capacity is fourteen and the gross weight 15,100 lb. A number of refinements over the Italian design have been introduced in the American model, probably the most noteworthy being the installation of contrast doors on the forward sides of the two hulls so that it is possible to enter from a single foot stage located between the hulls. The hatches of the former design are retained for the use of the crew, but the new doors are a great convenience for passengers.

For the first time in air show history the Fokker P-32, largest American built land plane, and the Consolidated Condor, largest American built flying boat, were exhibited under the same roof. This was not done without difficulty from the standpoint of the management and, in order to accomplish it, several portions of the seating space in the arena, as well as the entrance ways, had to be torn out.

Among the lighter planes exhibited on the lower floor of Madison Square Garden there were several new creations, most of which were of conventional design, the exception being the tandem plane displayed by the Ferris Aircraft Corp. This airplane, designated the "Crusader," has all of the characteristics of its predecessor the T-8, but is considerably smaller, being designed for two passengers, with a gross weight of 1,400 lb. The unusual features about the machine are its tandem wings and its landing gear which embodies the use of a nose wheel rather than a tail wheel. The Ferris might be considered as a conventional low wing monoplane with the addition of an auxiliary wing of smaller dimensions than the main wing, mounted at the extreme nose on a system of cable struts which raise it above the level of the main wing in flying position, the intention being that the front wing stalls rather than the main wing. The center of gravity is approximately at the leading edge of the main wing and considerably ahead of the two main landing wheels which have a 7 ft. track. The effect of this is that the plane is in normal flying position when standing on the ground and in landing. The power plant is the 4-cylinder inverted Kinner engine.

Among the more conventional craft in the exhibit were the EAC-1 introduced by the Engineers Aircraft Corp. and the Hestington "Governor" ship which was shown by the Hestington Aircraft Corp., both of these developments coming from Connecticut. The EAC-1 is a 2-place, externally braced monoplane, with anti-diving seating arrangement, in an open cockpit and powered by the Wright Gypsy engine. The plane has a span of 20 ft. and a gross weight of 1,400 lb. It is characterized by a sharp sweepback in the wing for purposes of stability and maneuverability of cockpit, and is provided with folding wing, permitting storage in a space of 20 ft. x



The "Governor" ship introduced at the show by the Hestington Aircraft Corporation

11 ft. The EAC-1 is of conventional construction and embodies a number of ingenious devices intended for comfort, probably the most noteworthy of these being the locking arrangement for the folding wings, which precludes the possibility of taking off with wings locked. An unusual feature is that no adjustment is provided in the stabilizer, but a device is included to take the tension off of the control stick or to provide a variable neutral position of the stick fore and aft.

The Hestington "Governor" ship is a cantilever cabin monoplane powered with the Warner Scarab engine and having side by side seating arrangement. Structurally and aerodynamically it is conventional. The span is 35 ft. 5 in., the gross weight 1,650 lb. and the wing loading 9.2 lb. One of the most noteworthy features of the design is the roomy cabin and the similarity to automobile characteristics. The stabilizer adjustment is controlled by a lever that bears a striking resemblance to the automobile emergency brake. The doors on either side of the cabin are rectangular in shape and of sufficient



One of the refinements in design was the seating arrangement in the cockpit of the Ferris monoplane

area to make entry and egress comparable to the present type of closed automobile. A design feature in this airplane that might well be emulated by other manufacturers of closed monoplanes is the transparent section of the wing behind the seat, making it possible to see above and behind. This feature was also noticed on the American Eagle monoplane exhibited at the show and on one or two others, but it is certainly worth while for designers to remember that a cabin airplane can not have too much visibility.

The Swallow Airplane Co. introduced its new Avulcan powered plane at the show. The new "Swallow Special" is an unusual span airplane, the upper wing being 31 ft. in span and the lower 23 ft. 2 in. The gross weight is 2,250 lb. The plane is a 3-place general purpose machine.



Interior showing controls in the Avulcan cabin. The small wheel above propeller is the big engine throttle control for the wingtip motor.

Having the structural characteristics of its predecessors, as in the case of the other new light planes at the show it was equipped with low pressure tires and an additional shock absorbing system in the landing gear. A number of refinements are embodied in the design, probably the most important of these being the increased luggage compartment space which in the present model is 30 in. x 10 in. x 12 in. and 8 in. x 34 in. x 18 in. and the provision for 75 lb. baggage weight.

One of the late arrivals at the show was the Taft



Photograph of the KAN-1 showing the slung wingpack of the wing.

Kingsley, a flying boat which is to be produced by the Whitcomb Manufacturing Co. which has purchased the capital stock and assets of the Taft Airplane Corp. Although the craft displayed at the show was a flying boat, it is to be produced as an amphibian under the name of Whitcomb amphibian. The machine has a wooden hull, wooden wing structure and fabric wing covering. It is of the pusher type powered with the LeBlond engine. The upper wing span is 37 ft. and the overall length 28 ft. 9 in.

THERE WAS A LOT of new things among the engine exhibits at the show. However, there were three new engines that were worthy of mention. Among those introduced were the Isotta Fraschini cross, and a great deal of interest was attracted by the 1,800 hp. 16-cylinder broad arrow type. This engine, incidentally, is liquid cooled and was developed by order of the Italian Military Authorities, from whom special patent had to be secured in order to exhibit it in this country. This power plant, which was by far the largest in amount exhibited at the show, is surprisingly compact. Several other interesting power units of the same make were also on display. The American engines introduced for the first time at the New York Show were the Beale Model K by the Knott Aircraft Corp. and "Pinto" by Messers Meers Inc. The Beale is a 2-cylinder radial type of conventional construction except for the cylinders which are of the U-tube type. The intake valve is located at the rear of the cylinder and the exhaust valve is located in a recessible cage at the extremity of the cylinder to facilitate cooling. The intake valve operates by direct contact with the taper, while the exhaust valve gear embodies a push rod and rocker arm. The engine has a bore of 4.5 in. and a stroke of 5.25 in. and is rated 135 hp. at 1,250 r.p.m. The weight dry is 370 lb.

The Messers Pinto is a 4-cylinder inverted inline six cooled engine rated at 95 hp. at 1,800 r.p.m. and having a weight of 270 lb. The engine is conventional in design and is distinguished by its simplicity of mechanism. Four gears inside of camshaft and magnetic timing and drive at three are of the same diameter. The bore is 4½ in., the stroke 5½ in. and the compression ratio 3:1. The length overall is 45½ in., the height overall 28½ in. and width 12½ in.

Included in the exhibit of the Cotten-Wright Co. was the new geared Cyclone engine, which is now being offered for commercial use by the Wright Aeronautical Corp. The engine is rated at 875 hp. Other Cotten and Wright engines also were shown.

As is the case of the Detroit Show, the exhibits of gliders lead one to believe that this branch of the industry is going to be important in the future. In addition to the Peil Glider had previously mentioned, engineless craft was shown by Waco Gliders, Detroit Aircraft and Franklin, the exhibit of the latter company being an "Eagle," the type used by Frank Hawks in his racing flight.

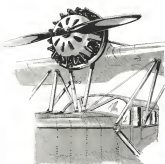
The adjustable rubber pulley of the KAN-1. An upward movement of the bar changes the position of the pulley to one of three adjustments.



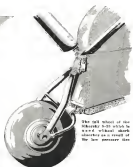
View from the rear of the KAN-1 showing the slung wing of the flexible tandem plane.



Locking device for the flexible wings of the KAN-1.



The flexible rubber pulley of the KAN-1. An upward movement of the bar changes the position of the pulley to one of three adjustments.



The rubber pulley of the KAN-1. An upward movement of the bar changes the position of the pulley to one of three adjustments.

Speed, THE INDUSTRY'S GREATEST SELLING POINT

Discussed by the S. A. E. and the A. S. M. E.
at New York, along with Engines,
Amphibious and other matters

G. H. Williams, whose paper on cylinder design and inspection was read

By HENRY O. PATTISON, JR.
Assistant Editor of AVIATION

A WIDE VARIETY of subjects beginning with aircraft engine types and continuing on to general discussions of the industry, offered much of value to the designer, manufacturer, or operator of aircraft at the New York show sessions of the S. A. E. and the A. S. M. E.

The subject of speed seemed to predominate in all sessions. Almost invariably, at each paper was read sessions was made that speed was aviation's greatest asset and must be developed. Types of engines and plans which did not foster and show promise of this development were considered useless. The speed element reached its climax with a paper by L. A. J. Williams.

George J. Mead, read the first paper, "In-Line Liquid-Cooled vs. Air-Cooled Engines." He started off with a delicate statement that his paper touched only on commercial applications of aircraft engines and did not deal with the military aspect in any manner. He stated that the initial effort in design of an aircraft power plant was to produce an engine light enough to permit flight. We are now faced with a complex issue of types and use which require very definite study. It was quite natural that much could be learned from the automobile engine.

The first air-cooled engines were of the rotary type using the rotary motion of the cylinder to cool them. Due to the centrifugal force, it was impossible to build an engine of over 200 hp. The general use of aluminum was made the liquid-cooled in-line and Vee type engine paramount. The Germans made several of this type during 1914. It was not long before it was discovered that a steel jacket could be put into a cast aluminum cylinder and this further reduced the weight of an engine and made higher power more practical. The necessity for greater power forced the reduction of weight so that by 1929 the Vee type of engine, water-cooled, was practically standard.

Starting with 1920, commercial operators evolved a need for greater reduction in power plant weight, as well as increased dependability. Before this the British had gained development of an air-cooled radial engine. During the last four or five years, this type of engine has been almost the standard power plant for our commercial flying. It is interesting to note that power plants of this type with approximately 500 hp are now being built with a weight of 1½ lb. per horsepower.

One of the main points made by Mr. Mead was that on an instance has a new type of power plant ever been accepted unless its weight per horsepower bettered or at least equalled that of its predecessors. While the weight per horsepower has been reduced, greater dependability has also been achieved.

The first and greatest requisite for aircraft engines is dependability. Next to this, comes minimum weight per horsepower.

Speed is one of the principal advantages of aviation. It is, therefore, natural to expect strenuous competition between the various projector lines in this regard. Thirty per cent of the available power is now used to overcome resistance which could be eliminated. At the present time, a 600-hp engine should give a speed that is now only possible with 800 hp. Large power plants, for big planes will be common in the future. It must be remembered, however, that power plants must be of the right size and rendered for such planes so that there may be a reserve supply, and that the plane may operate on 75% of the total power when cruising or in the event of failure of any of its engines.

Greater weight became common in order that the propeller speed may be reduced and the consequent noise and passenger discomfort eliminated.

The weight per horsepower of the air-cooled radial engine and the Vee type water-cooled engine, without a cooling system, are practically the same when engines of equal power are considered. The water-cooled engine adds 10% per horsepower for its water and radiator. The latest development in reducing this weight has been the use of ethylene glycol in place of water. However, when such a system is used it is no longer possible to cool the oil and it is therefore necessary to add an oil radiator. This adds drag and also weight.

It may be possible to reduce power plant drag to such an extent that a liquid-cooled engine of greater weight will more than compensate for this by making available a greater percentage of its power for the propulsion of the plane. At present, an air-cooled engine with a proper cooling gives an greater drag than the water-cooled type of engine. Much development may be expected in the reduction of drag in solitars. Mr. Mead thought it quite unfair to compare the fuel consumption of a Diesel cycle engine, operating in a compression ratio of 14 to 1 with a piston-type engine of the Otto cycle running at 5½ to 1. There is much room for improvement, and the greater compression ratio a great saving in fuel consumption should be observed.

In concluding, Mr. Mead stated that it is impossible

at present to predict which type of engine would be favored for the larger power units.

Captain Thayer of Langley Field, led the discussion with a statement that drugs might be so lowered that weight may be increased. He has high hopes for an engine which could be placed in the wing. Dr. Sanford A. Moss discussed superchargers in connection with engines.

He stated that at the present time, a Diesel may be supercharged to deliver 40 per cent extra power, while a gain of 25 per cent is considered good with the Otto type. There was much discussion on the two cycle engines. Many were in think that it offered a great future, and a Junkers engine, developed in Germany, was offered as an example. It seemed the opinion of the majority present that there was a great deal to be learned by research along these lines.

Mr. Erdikowich, of New York University, stated that reliability is two things. Important. Liquid-cooled engines will stand more punishment from bad fuel than will air-cooled radial engines. An interesting side-light on the paper, it was brought out that ethylene glycol has approximately the same flash point as gasoline.

THE SEVENTH PAPER of Tuesday afternoon was read by Mr. W. F. Davis and was entitled "In-Line Versus Radial Aircraft Engines." Mr. Davis started his paper by lamenting the number of engines which have been produced in this country without proper engineering development. The permitted, capital, and time required for the development of aircraft engines is very hard to procure. He thought that it was very well that the 1929 definition in the aircraft business had temporarily stopped the mad race to place in production anything and everything in the way of an aircraft engine that would fulfill the minimum requirements of the Department of Commerce. He thought that this had given the industry time for consideration of the pros and cons of various types of engines.

The main thing that aviation has to sell is speed. All other things are secondary, but not overwhelmingly so. Safety is paramount, and this depends almost entirely upon engine reliability. Escaping himself from any discussion of the Diesel type of engine, Mr. Davis continued with his subject. While reliability is the first consideration in the selection of an engine, it can only be proven by many service tests. Performance of accurate engines to date have not been adequate for the establishment of public faith in aviation which is essential to the proper growth of the industry. It is proper therefore to deduce that present-day designs are inadequate.

The inverted in-line air-cooled engines present features which cannot be duplicated in the radial type engine. The main feature is the ease of lubrication. It is much easier to make an in-line type of engine more rugged than a radial. Mr. Davis was also of the opinion that the in-line engine presents a simpler design and hence a more reliable type due to its overhead valve plan which eliminated the articulated rod design of the radial. The expansion of an exhaust valve on an overhead cam shaft arrangement adds rather than abates the engine's operation. A compressed radial valve gear tends to correct this condition, but does so at the cost of complications and reduction of reliability.

The simplicity of air-cooled engines itself for reliability regardless of the type of engine. The use of a supplementary air-cooled cylinder is definitely limited by the cylinder-cooling capacity. For reliability's sake, therefore, it is well to employ as a minimum cylinder size one well within the limits of satisfactory cooling. Power may then be increased by increasing the number of cylinders. On a radial engine the number of cylinders is limited by the push and assembly. Mr. Davis thought, however, that it was not impossible that six, twelve, or even twenty-four cylinder in-line engines might be developed. It is possible to adequately cool any number of cylinders of an in-line arrangement.

The radial engine can be made lighter in weight for a given cylinder displacement. However, the limitations placed on its practicable expansion speed due to the load of the heavy connecting connection and assembly and the high inertia of the push rod type valve gear make it really possible for the in-line engine to equal or better the specific weight per horsepower by increased output and a high speed with no sacrifice of reliability. This is due to the low crankshaft and bearing load in the in-line type of engine.

The in-line inverted engine with completely isolated valve gear is a much cleaner and quieter engine than any other so far developed. Improved visibility is an obvious feature of the inverted in-line engine. It also permits a lower landing gear. It costs somewhat less to produce radial engines than in-line engines. This loss, however, would be completely offset in quantity production by experience gained from the automobile manufacturers.

It costs slightly more to mount an in-line engine than a radial. There is very little difference, however, between the total cost of the installation of both engines when they are completely cooled.

Mr. Davis was of the opinion that, as to service costs, the life of the high-grade in-line engine would greatly



superior than the radial. The in-line would go much longer between periods of overhaul than the radial. He was quite concerned that the in-line type of engine suffered a solution for many of the present problems of the radial engine.

The discussion on this paper centered mostly around the frontal area and resistance of radial and in-line engines. Captain Brewer, of Langley Field, brought out the major point when he asked the designers that were present at Dayton and Pitts for more cooperation between designers and engine builders to the extent that drag might be eliminated and interference greatly cut down. He stated that at present an engine is designed to have a very low frontal area and little drag, and in mounting the engine in a plane so much interference is created by wings and other members that the entire benefit of the engine design is lost.

Mr. Allison of the Navy Department, was of the opinion that the drag of a radial engine could be reduced to the equal of an in-line engine by good cooling.

Col. V. E. Clark, who was to have presided over the afternoon session was unavoidably delayed and his place was, ably filled by Mr. C. H. Hoffmann of New York University.

The first paper of "Today afternoon" was "Amphibian Design and Transportation," by G. M. Bellanca. Due to Mr. Bellanca's absence this paper was read by Mr. R. M. Mock. The author prefaced his paper with a short history of amphibian development.

He stated that the public has long desired amphibian planes. It is easy to ascertain the demand of the public, but it is not so easy to satisfy this demand. Unfortunately, he believes we do not as yet know how to make such that will completely satisfy the demands of the public. The difficulties are numerous in the design of an aircraft are weight and resistance. This difficulty exists more in airplanes than biplanes, and still more in amphibians. He cited as an illustration of this point the Bellanca Parawater, showing that it might be the closest to a land plane, having some of its efficiency when made into a seaplane, and becoming almost impractical when made into an amphibian.

A hasty investigation made by Mr. Bellanca of the weight percentage of the landing gear in a seaplane or a flying boat showed it to vary from 5 to 7 per cent of the gross weight, or about 30 per cent of the dry load. In the case of an exposed landing gear he found that the speed was reduced approximately 0.6 per cent. In the face of such difficulties it is not surprising that the development of amphibians has been slow.

Mr. Bellanca was very emphatic in his statement that an amphibian today which was successful could not be merely a matter of time. Any plane designed to land on water has a large percentage of its power absorbed in resistance alone. Added to this the weight of such a hull or float makes the problem extremely difficult. He was equally emphatic in his statement that in the face of all these difficulties the amphibians which are now on



1. That G. M. Bellanca is the originator of the Dux-Duc.

the market are remarkable examples of good design. They need, however, be improved.

Amphibian transportation has advantages which cannot be overestimated. It makes possible, on many occasions, the saving of large amounts of time which are now occupied in going to and from airports at the larger cities. To offset this, however, there is the fact that amphibians at the present time are slower than land planes.

Concluding the paper, Mr. Bellanca made it clear that the amphibian must carry a sufficient payload and be endowed with sufficient performance to compete with other types of aircraft. His closing sentence was "Proper design is essential."

Discussion on this paper started off with a statement that amphibians used in the United States is more limited than in five years. While many claim that the planes are safer because of the frequent bodies of water it is in many ways difficult to land them except in a very wide stream and where there is little, if any current. In addition, their price is very high.

It was the general consensus of opinion that amphibians were useful and would always find a market with small private owners.

Someone from Fokker company stated that they had had much trouble in getting the biplane on amphibian wheels in fraction of a mile. It was thought that the Sikorsky company had eliminated this difficulty with a little of its own design.

When all advantages and disadvantages of both types were added, the final consensus of opinion seemed to favor the hull runner but the problem type of amphibian, that is the conversion of a flying boat rather than a seaplane into an amphibian.

The second paper of the afternoon was read by Mr. J. C. Housholder of the Goodyear Zeppelin Corporation and was entitled "Transoceanic Air Travel." The first point brought out by the paper was that there is no law of transoceanic flying have not been of great help as far as statistical conditions are considered. All of these flights have been made with overlanded planes and without landings.

An airplane can fly the Atlantic, providing it has all the elements, both mechanical and physical, in its favor. A large enough flying boat might be able to make a forced landing and rely on reasonably high seas. In general, however, a forced landing on the high seas cannot be tolerated by commercial enterprise.

Large airplanes are being built, but their endurance unfortunately is not increasing. The spectre of engine failure is serious for a dynamically supported aircraft. Multi-engine airplanes reduce this danger somewhat, but multiply the chance of engine failure. Mr. Housholder seemed very definite in his point that there is little utility to gain from multiple power plants. Airplane or flying boat as known and built today cannot be made safe enough for transoceanic air travel.

The large airship, on the other hand, are ideally adapted for such work. Repairs may be made in the air and the failure of any of its power plants does not make disaster. It cited as an example the Zeppelin flights across the Atlantic as having been successful except one, where a forced landing was necessary. In this case he made a point of the fact that with four out of five engines dead one, engine brought the ship safely in.

The freedom of an airship for dependence on any single piece of its structure or machinery is in direct

contrast to the airplane and therein lies the big advantage of an airship. On the ground however the airship is only safe when securely grounded or beached. Here the airship is very weak compared with an airplane. He compared the airship with great ocean liners, which are safe enough at sea but can only enter several ports of the world, even then they are in danger of grounding and at great expense from pier charges and tugs.

All of these points were directed toward proving that airplanes and airships are not competitive, each having its individual field.

Airships are relatively inefficient in smaller sizes, hence the commercial airship will always be large craft with an extremely large initial cost.

His next point in the paper was the comparison to distance and weather conditions of all the man-made means. Mr. Housholder stated that even though artificial islands were anchored across the Atlantic there expense would be too great for practical operation. As lanes of the future must be able to fly on regular schedules over regular routes and take the weather as it comes, he is good or bad. In this connection the Zeppelin has a decided advantage since it may ride out a storm or, due to its high speed, may maneuver to avoid it entirely.

Whereas the airplane, when started, must continue on to its destination without delay, the airship may delay its time of arrival indefinitely to wait the weather conditions.

At the conclusion of this paper, Major Hoffmann introduced Col. James Patterson, who started the discussion. His main point was that, if the airships did leave to without passing through storms, they would be lowered, and speed is the only factor which the aircraft industry has to sell at present. Comfort and other things cannot be offered to the public until this can be proved in other transportation media. The discussion on this paper was short and everyone seemed well satisfied with its presentation.

Mr. Housholder presented approximately a dozen lantern slides to illustrate his weather signs and the aerodynamic and engine mounting details of the new Goodyear Zeppelin.

WORMSLEY'S warning was given over to the discussion and the making of papers on two very interesting aircraft. The first of these was entitled "Do-X Flying Ship" and was written by Dr. Claude Dornier of Germany. The paper was presented to the meeting by Dr. C. H. Schildknecht of the Dornier Company of Augsburg, Germany. So much has already been written on the Do-X that it is necessary to entirely abstract this paper here. A description of the plane was published in the January 4, 1938, issue of AVIATION.

There was no discussion on Dr. Dornier's paper but it was followed by a very vivid set of pictures showing the stage ship in construction and showing its test flight.

One interesting feature which has not previously been covered was the chart prepared by Dr. Dornier showing routes over which the Do-X would be protected, and giving

the number of passengers which might be carried. Much interest was manifested in the cost of operation of a plane of this size. Dr. Dornier's paper showed that it operated at a cost of from \$2.78 to \$6.94 per mile, according to the distance flown.

In concluding his paper, Dr. Dornier gave out a few general statements of the aviation industry. He did not think that the industry was as far advanced as is generally thought. He thought that it is greatly hurt by the pessimistic people who say that those who build planes of the Do-X type are in advance of their times and are unreasonable. The Doctor said that he is positive that he demands for things costs.



2. That J. C. Housholder who discussed speed in his paper.

This paper opened with a short review of the history of the Zeppelin. The experimental work began in 1919 and involved a number of unsuccessful attempts to create a satisfactory lift system by means of two separate lifts resulting in opposite directions. It was not until a single lift system was developed that the Zeppelin was launched freely to the trials that flight was successful.

Mr. La Page pointed out the fact that there were no fundamentally new aerodynamic principles involved in the design of the Zeppelin, there being merely a new application of these principles. Because the details of the construction of the Zeppelin are not in the public domain, Mr. La Page confined his discussion to the simpler aspects of the mathematical relations involved. He went to sufficient length to show clearly the mathematical basis for the lift and lifting effects of the mast lift system. The theoretical conditions concerning a mast lift system in rotation only are considerably more complicated by the introduction of a motion of translation of the entire system. This satisfactory solution of this aspect of the problem was presented out to be the outstanding characteristic of the interpretation of the Zeppelin.

The paper then discussed the effects of the longed-for blades into the rotor hub. The blades are hinged internally and are re-acted when at rest by the action of a system of rubber cords which press them then from aerodynamic pressure or other parts of the machine. When in rotation centrifugal force maintains them in the flying position and the load is entirely removed from the supporting cords. It is stated that in an airship of above 2,000 lb weight the centrifugal force on each blade would be in the neighborhood of 5,000 lb at 180 r.p.m., making the inertia forces in each blade about ten times the lift forces.

The stresses on the blades are therefore of a considerable different nature from those encountered in an ordinary airplane wing. On account of the necessary rigidity of the airship the wings have been hinged in a vertical direction to permit a small amount of relative motion between them and to relieve the blade hubs of unbalanced

drag forces which change periodically in the course of the cycle. Inter-ladder bracing has been introduced to take care of the unbalanced forces.

After leaving the discussion of the rotating system Mr. La Page described the auxiliary parts of the autogiro, which were very similar to corresponding parts of the conventional airplane. It was pointed out that, contrary to airplane requirements, a tail surface producing lift is required to offset the positive moment which tends to make the machine too heavy in flight because, due to the high mounting of the rotor, the point of intersection of the lift vector and the center line of the craft moves forward of the center of gravity as the speed increases. The reasons for the use of the short wing were discussed and also several of the usual characteristics such as inclination of the thrust line, etc.

THE AFTERNOON came two general papers, the first of which was "Speed Things" by Lutz Alford Jr., Williams, Jr. In his paper Lieutenant Williams repeatedly stressed the fact that commercial aviation must speed up if it is to get anywhere. At the present time it is far too slow to compete with well organized railroads.

One of the major points brought out by Lieutenant Williams was that as it is impossible to predict what speed will be reached by airplanes. Each time a new record has been established everyone has said that it approached the ultimate level, but it is always increased soon after.

Lieutenant Williams paraphrased much of his racing experience for the benefit of those at the session and told of many humorous incidents as a result of flying racing engines. He concluded his paper by laying great emphasis on the fact that the younger generation is growing up to accept aviation as a matter of course and high speed as commonplace. It was this fact, he held, that delayed much of the progress so often found.

THE closing paper of the S.A.E. session was delivered by Major Leslie MacDill of the Army Air Corps and was entitled "Some Aspects of Commercial Aviation from the Viewpoint of an Army Officer." The paper touched in general on all phases of commercial aviation and detailed in particular the need of advancing the military, naval, enterprise and industries to needs of the country in time of war.

Major MacDill made several vital points during his paper. One of these was the fact that the total profit on all government contracts could amount to no more than \$1,000,000 per year at the present budget levels. This \$1,000,000 must be divided among all countries holding government contracts, hence it is easy to see the approximate amount of profit and earnings to be made by these companies.

No figures were available, he said, to indicate the total amount which might be earned by commercial aviation, but it was not greatly in excess of the amount of government contracts. Thus he said it was easy to see that with all the companies which are now manufacturing planes and accessories the earnings per company would be very small.

One of the major points of this paper was the decreasing of the fact that large banking interests had entered aviation. Unless a company shows a substantial return on the money invested bankers are not willing to foster it. Commercial aviation cannot at present show these earnings and must be largely a matter of faith. It is always possible to have a direct express transport plane more reliable than a single engine plane, but Major

MacDill did not believe that the American public will pay for the added cost associated. He was quite firm on the point that the single engine transport would be the plane of the future.

In a discussion on this paper the main point brought out was that Americans are inclined to accept foreign ideas. On the other hand, foreigners really accept our innovation that we develop. Mr. W. F. Allen of the General Electric Co. said that he did not believe that the Europeans could produce better engines and planes than we could but that he believed they would do so sooner because of this cause.

WEDNESDAY EVENING the Metropolitan Section of the S.A.E. held its annual dinner at the Park Central Hotel. At this time a paper "Transportation Versus Air Cities" was read by Mr. F. H. Reinhardt, President of the Aeronautical Chamber of Commerce of America. The paper was presented in a good manner and was very well received.

THREE PAPERS were presented at the session held by the S.A.E. on Thursday. The most extensive of these was that by C. H. Biddlecombe entitled "Service for Airplane Engineers." This paper gave quite a detailed report on the different service ideas employed by the various engine manufacturers. Mr. Biddlecombe presented in great detail the trends of industry along this line and the present-day systems as employed by the larger engine manufacturers. It was his opinion that while the engine manufacturers of the industry continued to give the service which they do now, or for that matter, better service, the growth of the industry will be considerably hampered.

Another paper, "Engine Problems in Aircraft Servicing," by George Drysdale of Brooklyn, New York, went into the engineering side of the service and discussed its own systems. The paper's main points and details seemed to be a plea for more co-operation between the engineering departments of the airplane manufacturers and the servicing departments. By proper co-ordination of design the service may be greatly aided and the cost much lowered.

Mr. Drysdale believed that airplane manufacturers would find it expedient to adopt the policy found in the automotive industry of a time-out repair schedule.

The third paper on the subject of aircraft servicing looked at it from the point of view of the owner. In his paper, "Airplane Servicing from the Operator's Point of View," Richard C. Marshall of the Thompson Aircraft Corporation said that the growth of aircraft servicing had been slow or less without control. At the present time there is absolutely no standardization in this structure, and many opportunities are lost for economy.

The smaller repair shops and individual operators cannot effect the economies that a large repair shop with a constant stream of business can. For this reason Mr. Marshall believed that the large deposits will be the thing in the future. He cited as an example the Thompson Aircraft Corporation's services, showing how they have consolidated their work and employed the best type of workmen to effect economy.

Mr. Marshall then went into such detail as to the individual operations of repair, showing how in every instance they had anticipated and actually accomplished in improving the processes actually employed by the manufacturers.

CREATIVE Wing Design

The Possibilities of a System for Coordinating Design Procedure

By RALPH H. UPSON

THIS INVITATION from Mr. Warner to expand further some of the material from my recent S.A.E. paper "Wings—A Co-ordinated System of Basic Design," S.A.E. Journal, January, 1959, sufficiently indicates the interest in the subject. Yet I had hoped that further commitments would be from others, because the principal object of the paper referred to was not to set up any new rigid system of design but to show by practical example the possibilities and importance of some system for co-ordinating and focusing design procedure. There is probably no serious lack of system in the ordinary routine of checking and running a set of pre-assigned ideas, but there is a comparatively neglected field for system in the creative part of the design—for picking the ideas themselves. And here perhaps the most fundamental need is an adequate system for establishing the best general proportions and arrangement of wings.

The main trouble seems to be due to a kind of hang-over of bad habits from the pre-World War days, though I doubt if there is any completely accurate engineer today who doesn't use the principle of induced drag.

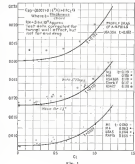
Now what are some of these lingering, though obsolete, habits that don't jibe with the new principles? First, is the method of picking an airfoil, from the many hundreds available, by direct comparison of their span loaded characteristics. Apparently the old idea was that some day all these scattered results could be brought together into a monolithic volume, something like a mail order catalog, where each little airfoil would be illustrated and its particular merits set forth. We might imagine a master collection of tests on the weight, strength, and deformation of model wing structures possessing more or less haphazard differences in airfoil shape, arrangement, materials, and scale effect, the latter mostly unknown. Indirectly, such a book would be very instructive; but its use for direct design purposes would be profitable only in a rather primitive crivo-ma state of the art.

CONTINUING the analogy of the store wing structure a logical design will proceed directly from a consideration of the forces that have to be carried, with due regard for the limitations of span, materials, etc. The same applies to the aerodynamic aspect of wing design. In fact the most distinguishing mark of modern aerodynamic design is its emphasis on the forces themselves and their direct interrelations rather than on the so-called "airfoil characteristics."

And here we are almost on the point of tripping over

another bad habit, inherited from the old school, that of dealing with the airfoil as something more or less distinct from the wing structure. Someone will probably rise to protest that requisite span depth and spacing is added to in all the time honored text books. *Quare vero*—but more attention is not enough for a studentship to dose that for certain purposes it almost eliminates the identity of the parts. If it weren't for the structural requirements there would be no aerodynamic problems, because the principal sign posts of efficiency would all point in the same direction. Similarly if it weren't for the aerodynamic requirements there would be no structural problems. In other words, the two problems are combined into one in the preliminary or creative design. And if this part of the design is wrong, the design as a whole is wrong and little or nothing can be done later to "fix" it.

In recognition of the above relationship, the following semi-empirical equation for "structural efforts," gives the direct relation between the aerodynamic forces and



the structural proportions, without reference to the "airfoil section" as an arbitrary model.

$$C_D = \frac{C_D^0(1+k)}{\pi} + (10.007 + 0.1V^2)(1 + 0.81V^2)$$

where k is the aspect ratio (span/mean chord)

V is the velocity ratio (Mach no./0.3)

$(1+k)$ is plus from factor (usually 1.02 to 1.06)

C_D^0 and C_T are drag and lift coefficients (N/V² A² coefficient)

THE FIRST TERM above is of course the induced drag and the second the profile drag. The latter is for a Reynolds number of about 3,800,000, and is intended to give the diskier sections an estimated drag value corresponding to laminar flow. Fig. 1 shows its relation to a series of tests on actual airfoils with squared tips.

Several other airfoil formulas have recently been suggested—one of them derived from tests in the Variable Density Tunnel, being superficially similar. But all of them, as far as I know, try to do too much by including the effect of detail section changes, which have no direct connection with the structural problem, and therefore have no place in the basic formula. The mechanic counter, for example, is only a direct means to an end—value of refining the aerodynamic properties inherent in the basic proportions—and one which should be already known from a consideration of the approximate plan form and spinning range of lift coefficient. In the same way the detail arrangement of internal structural parts is a means to an end—and one which the competent engineer knows in advance from the general external proportions alone. To properly include such incidents in a single basic formula is quite impossible.

As a matter of fact the formula just given is *quite* only insofar as it relates to the aerodynamic analysis of those basic wing proportions which have both a structural and aerodynamic significance. To complete even the preliminary picture we must work back through these same proportions to the weight, which is the general structural loading in a similar way that the drag is the principal aerodynamic loading. Telling part of the structure out of the wing and calling it "parasitic" doesn't change the principle of structural loading, but

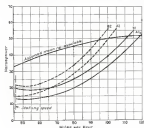


FIG. 1

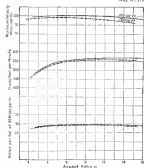


FIG. 2

only complicates it by putting part of it to one side as a drag effect.

And then we soon find that besides these back door and side door connections between weight and drag, there is the still more important connection from drag connection. For although the drag has little or no direct effect on weight, the weight has a considerable effect on drag, especially at low speeds, and both are of course, factors in various important performance characteristics. These fundamental quantities, not to say wing design purposes alone, must hence be expressed as properties of the entire airplane, with allowance for interference and other secondary, though important, effects.

The usual method of achieving this necessary coordination consists of testing in the wind-tunnel a model of some or less preselected proportions, for which the weight effects are independently estimated and weighed, together with the wind-tunnel data, in computing the full scale results. Such a method, intelligently followed, is of value at best in a negative way. It is a good check against the possibility that wind-tunnel will seriously increase the drag or injure the stability and control, usually in a static sense. It would also be a direct means of forecasting performance if it weren't for the discrepancies introduced by scale effect, air-stream and small passage parts. A note of scale experience, particularly in the use of some one tunnel and subsequent flight tests, will often be able to make surprisingly accurate corrections for these discrepancies, especially when they tend to counteract each other. But the corrections are arithmetic, and of a smaller range of magnitude to the test results themselves, and the same experience, further mentioned, should be capable of forecasting the final results without the continued aid of the wind tunnel, except in the matter of interference and special cases of turbulent flow. Thus the wind tunnel, though sometimes an important accessory, is now only an accessory, and should never

be regarded as the central test around which practical design procedure revolves.

Scientific research in aerodynamics is of course very much dependent on the wind tunnel or its equivalent, but its value depends on being able to test a whole series or family of related models, and even here the ordinary aerodynamic tunnel has apparently passed its peak of greatest usefulness. It is not only that the wind tunnel is limited to the aerodynamic side of the problem, but by its very nature it focuses attention on preselected arrangements and proportions which may be all wrong—and more than likely are all wrong, unless the design has already been established by a more fundamental and direct method.

SUCH A METHOD must involve a combination of weight and drag for the complete airplane in terms of the various basic proportions, and in a form convenient for manipulation. It has been proposed, it takes the form of two equations, one for weight and the other for drag. The latter is developed primarily from the airfoil equation already given but is modified to include the tail surface interference effects, etc. The weight equation simply shows the direct structural effect of changes in proportion on the structure of small and medium sized planes. For both weight and drag, the distribution of component forces is not primarily with respect to the specific part of the plane involved, but rather with respect to the manner in which the different items vary. The equations follow in a form in which the aerodynamic coefficients as used in this particular study.

Symbols of symbols (all dimensions in feet)

b = total span

c = root chord (at rear of cowl)

a = root thickness (at rear of cowl)

$(m-a)$ = effective structural thickness

t = $\frac{a}{c}$ = thickness ratio (constant for any one plane)

x = distance of lateral center of area from root = $2(A-2)$ for rectangular wing-braced hull wing set

$128(b-2)$ for rectangular wing-braced hull wing set

S = net planform wing area

S_1 = static area (area cut off in freestream)

$\frac{S_1}{S} = 2 + \frac{a}{c}$

δ = cowl area ratio = $\frac{S}{S_1}$ for cowl area wings or $\frac{1}{2}$ for externally braced wings

π = aspect ratio = $\frac{b^2}{S_1}$

Given M and $B = M + 1 - \pi$ or $\frac{M}{\pi} + \frac{M}{\pi} + \frac{B}{\pi}$

$1000 + M' + 0.01T + \frac{0.28V}{S(x-0.2)}$

$1.25V\delta\sqrt{\frac{1}{\pi} + \frac{1}{\pi^2}}$

where B' is the "fixed loss" comprising all that is not a direct function of the wing proportions

M' is the weight of outside wing structure area for "cowl" wings or 50 lbs. (estimated) for the externally braced examples

B' is the covering (here plywood) of wings and tail surfaces, and various fixtures

π' comprises the wing structure and fittings

B' comprises wing ribs (plywood) and reinforcing blocks

B' comprises rib webs or lattices

W_1 is the wing torque loading ("wing twisting"), here assumed carried by the plywood skin

Total drag at 100 m.p.h.

$D = D_1 + D_2 + D_3 + D_4 + D_5 + D_6 + D_7$

$= D_1 + D_2 + 0.225 + 3.2\delta + 0.0 + \frac{0.014V^2}{\pi} + 0.00004(D_1 + D_2)\frac{B'}{W_1}$

where D_1 is that part of the direct parasite drag which is unaffected by changes in the wing tail surfaces at their external structure, includes direct interference

D_2 is the drag of the external wing and tail bracing, zero for cowl wings or 30 lbs. (estimated) for the externally braced examples

D_3 is the skin friction portion of the profile drag including tail

D_4 is the added profile drag due to thickness ratio

D_5 is the added parasite drag (due to true angle of attack) here assumed zero

D_6 is the induced drag, including induced moment, the latter taken at the maximum of 12 per cent

D_7 is the added profile drag due to true angle of attack as measured by the lift coefficient

The drag for speeds other than 100 m.p.h. is obtained in simple proportion. The first four terms vary as V^2 , D_5 is constant throughout the speed range, D_6 varies inversely as V^2 , D_7 varies inversely as V^3 . Three or

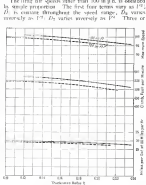


FIG. 3

four different speeds, properly chosen, are usually enough to establish the required drag and power curves. The coefficients in the weight equations will of course change with different types of structure and materials. For other than the assumed conditions, the drag equation

In addition to the Lumborgh Beacon on top of the City Hall tower, lights at Grand Central Air Terminal, Los Angeles International Airport, Chino Field and Long Beach Municipal Airport are at the revolving type, while the beacon at the Los Angeles Airport is a 360 deg. flashing type, which flashes L.A. in the International Morse Code.

Crossed Flares. Santa Monica, with a night runway 500x2,200 ft, is the dilemma of an almost unending need and is supplied with first one bank of ten Crosse-Hinds floodlights, located on the southeast corner of the field, and so directed as to illuminate the entire landing area. This installation has proved popular with pilots because all landings are made over the bank of lights and thus is no glare. Night visual lights are being regularly continued from Chino Field by Golden State Airways, operating a Ford transport company.

The Elmer Field Airport, located seven miles east of Los Angeles on Whittier Blvd., is a main runway, 5400 feet overall and opened by Roy Patton and Avem Black. Two runways, each 600x2,000 ft., extend in S.-E. & W.-N.W. & N.E.-S.E. directions, and are satisfactorily lighted for night flying by ten Crosse-Hinds floodlights. Field obstruction and runway marking lights are also installed.

With two intersecting runways, each 1,000x4,000 ft. and extending approximately north and south, east and west, the Los Angeles International Airport has complicated night flying with the aid of a single 180 deg. H.B.T. floodlight mounted at the southeast corner of the runway intersection. The floodlight is directed almost ender runway as the wind changes and landings are made along the light. Since the runways are perfectly smooth, this illumination has been found very satisfactory, although additional lighting is planned as night traffic increases. Revolving lanterns, rotating light, runway markers, and obstruction lights are also installed.

The Long Beach Municipal Airport, with a single east-west runway in use at the present time and a proposed west wind, was the first airport in the Los Angeles metropolitan area to be properly lighted for night operations. This field has been equipped for more than a year with revolving beacon, obstruction light, runway markers, and field floodlights and has been equipped with a gas flame ground beacon marking the letters L.B. at the west end of the runway. These letters serve to identify the field and have proved ac-



Showing the illumination of field on the main runway at the Los Angeles airport.

tremely easy to pick out from among other ground lights because of the flashing yellow gas flames.

Because of night operations over three different runways converging at the Western Air Depot terminal, officials of the company found it necessary to make an installation of lighting equipment, which will be greatly amplified as the airport is further developed. Approved recently \$81,000 has been expended on lighting to date with installation by the McNabb Electric Co. under the supervision of Thomas C. Nix, chief electrical engineer for Western Air Express. The W.A.E. field at Vanadena will eventually include surfaced runways in three directions, but due to the presence of a windmill



A floodlight approach light on the Grand Central Air Terminal.

will be more than 50 per cent of the time, this run way has been the first to be completely surfaced and lighted. Equipment now in use includes a wide boundary lights and green runway lights around the 500x2,200 ft. runway with red obstruction lights on all power line poles. A single 50,000-cp. arc light floods the field from the head of the runway so that all planes land over the light source. Passenger planes, such as airlines, regularly at this field either from San Francisco, Salt Lake City, Kansas City, and San Diego, while the Los Angeles Salt Lake coming at 2:00 p.m. on the night schedule.

In addition to these scheduled operations night visual flights are conducted each evening and have proved extremely popular. Lighting of the Los Angeles Airport was completed in August to give this field the first A-1 rating on the Pacific Coast for lighting. With a spare unit of available lighting, the field will eventually be provided with five vertical runways. Now, since the prevailing wind is usually the only runway yet surfaced and floodlighted is the east-west runway on the south side of the field named the administration building and hangars. Red obstruction lights are mounted on every power line pole

or obstruction within 50 feet to one gliding angle of the field in any direction. The landing area is completely outlined by white boundary lights, with green marker lights at each end of the runway in use. Seven Crosse-Hinds floodlights, mounted in a single bank, located just west of the administration building at about the center and on the north side of the runway in use, give excellent illumination for all night operations. The administration tower is illuminated by a powerful flashing beacon, while the upper portion of the tower beacon

the beacon is lighted on all sides by vertical red neon tubes. On the roof of the tower hangs three arc spot lamps, one an arrow pointing north with the letter N in the center of the arrow, and the other outlining the letters L.A. as a further identification of the field. All buildings on the field are completely lighted on all sides and top during the hours of darkness. Lighting equipment was installed by the Los Angeles City Bureau of Public Light, under the direction of C. N. McCauley, and no effort was spared to make every part of the installation permanent in character. A Crosse-Hinds ceiling light is installed in the field's lighting equipment and special attention has been given in the lighting of hangar interiors so that, by the use of a light reflecting surface, oblique lights, shadows have been almost completely eliminated around the planes.

WITHIN SERVICE of the Grand Central Air Terminal, it was necessary to install night lighting equipment to call attention to the current local illumination equipment issued by the Department of Commerce Inspectors, and electrical suggestions from pilots using the field in order to be sure that the resulting system would be the most complete which could be laid out. Conditions at the Grand Central Air Terminal presented a number of real problems to adequately lighting. The field has two runways running almost at right angles to each other, and the wind shifts so that it was necessary to provide for landings in either direction on either runway. In addition to this the field is located either in the downtown district than any other Los Angeles airport and is consequently bordered by industrial, commercial and business districts, which make it doubly important that the field runways be amply illuminated in order that the pilot may easily distinguish his landing area from the mass of surrounding lights.

After considering bids from various companies, the lighting contract was awarded to the Crosse-Hinds Company, of Syracuse, N. Y., and all the work of designing the complete electrical installation was turned over to Homer Whitely, head Crosse-Hinds manager and consulting engineer. With the exception of the installation and control panel in the tower control room, which is of Woodhouse manufacture, all equipment used at the Los Angeles airport, with several exceptions furnished by the Crosse-Hinds Co., with several exceptions furnished by the Stupker Electric Co., of Burbank, Calif.

The field is marked by a 24 in. revolving beacon mounted on top of the depot tower 61 ft. above the



One of the beam-outlined areas of the Los Angeles airport. This area is on the east of the main runway and indicates north.

field level. This beacon is of 7,000,000 cps. furnished by an incandescent lamp equipped with a quartz light which is automatically switched into place if one light burns out. Beneath the beacon are located two green course lights, one on each side of the beacon on a north and south line and each flashing in the International Morse Code the letters G.C. (Grand Central), plus after the beacon, rotating at 6 r.p.m., shows the course light. Night wind direction is indicated by three illuminated wind cones and one T. The cones are yellow mounted above black crosses, two being located on the hangars and one at the west end of the auxiliary runway. The "T" is mounted on a pole at the head of the main runway and resembles a monophone with a wing-shaped of approximately 20 ft. Green lights in the form of a "T" serve to indicate the wind direction.

Boundary lights are installed on a series loop extending around the field with lights and more than 250 ft. apart. These boundary lights are located just inside the fence and with the loop at the same elevation as the fence top. An H.I. transformer is connected in the main concrete base at the base of each light, and is provided with a series fire cannot socket so that if one light fails the current short across and consumes no current. Current is carried to boundary lights by No. 8 single conductor 2,900 volt non-insulative cable buried in the ground. All boundary lights are white with frosted globes.

Glare approach lights are placed three at each end of the main runway and two at each end of the auxiliary runways. These are of the top-down type in order to minimize damage if struck by a plane.

Red obstruction lights are placed on the roofs of hangars and all buildings on the field, on all poles adjacent to the field, and on every object within a beam to one gliding angle of the landing area.

All airport buildings are floodlighted to give at least two and a half foot candles of illumination into all portions of aisles and roof and wherever roof markings are used the illumination is increased to 15 ft. candles. Ordinary white light is used on all buildings except the depot, which is illuminated through amber lenses. There are 60,500 watt lamps used to light the depot, while 40,500 watt lamps and 35,500 watt units are used on the hangar markings. These are arranged on standard non-standard curves either one, two or four 1,500 watt units and one 500 watt unit according to the location on the roof. Side lighting is supplied in a total

of 189-200 watt angle reflector units mounted on 15 ft. cantilevers along the edge of the roof and set feet out from the building side wall.

A 14 in. collimating searchlight is mounted on the highest roof 600 ft. from the depot control tower, the light being being elevated at an angle of 63 degrees 25 minutes to the horizon. A collimating indicator calibrated in feet is mounted on the parapet of the depot tower, just outside of the control room, and it is only necessary for the operator to switch on the light and step to the indicator to read directly the height of the ceiling.

FOUR BANKS of floodlights are so placed that it is possible to land in either direction on either runway and always land over and away from the center of light. Although but 15 ft. wide, the runway is so placed that the landing surface thus securely is far exceeded everywhere as an added safety factor in the operation of transport planes at night. A total of 19,300 watt 32 watt type DCH-24 Crome-Fluor Flood lights equipped with 24 in. premium mirrors are used in the four banks. Six lamps are used on bank No. One at the north end of the main runway, five lights are used in bank No. Two, located south of the grassland and near the intersection of the two runways in a position to help light the main runway for north to south operations or the auxiliary runway for east to west landings. Six projectors, one used in each bank, are mounted on the roof. No. Four, located respectively at the east and west ends of the auxiliary runway. Bank No. Three serves to light the main runway for south to north operations and to assist in lighting the cross runway for east to west landings, while the fourth bank of lights serves to illuminate the auxiliary runway for north to east landings but can be used for increasing the illumination across the main runway if this should be desired. Thus it is possible to use three of the four banks and 17 of the 19 individual light projectors in lighting the main runway for its normal use as north to south take offs and

Strips of the floodlights are equipped with 40 degree lenses, and the other three, located at the runway intersections, with 80 degree spread lenses. Unseen and "quality" lighting of the runways is eliminated by locating the lights in banks and so directing each beam that there is a 50 per cent overlap with the next bank. These beams are focused so that all direct light is confined to well seen feet above the runway and practically no spill light is cast above that height.

In addition to the field floodlights there are three auxiliary flood lights mounted on top of the grassland strip just south of the runway. These lights are aimed and the beam directly in front of them for night stops and progress which are regularly staged. Turning before and after landing is facilitated by three night stop area lights in front of the main depot, one of 1,500 watt capacity and two 1,000 watt lights. These serve to illuminate the landing and taxiing platforms and after a plane has landed by means of the field floodlights the latter are cut out and the area lights switched on, making it much easier to taxi up for taxiing or departing of passengers.

Electric current for the lighting system is supplied and controlled independently of any other electric installation in the facility. A main switch at the electric substation and also completely underground, is located just north of the depot and control tower. Into this

switch is directed the power line of 2,300 volts and from it 64 control wires lead out to a small dark control panel in the control room of the field tower. This control panel was designed by Horace White, the engineer in charge of developing the lighting system, and was constructed by the Westinghouse company. There are 12 circuits embracing every phase of the night lighting installation, which are remotely controlled from the dark panel in the glass enclosed control room. Pilot lights opposite each switch on the control panel are an added guarantee of positive control of any portion of the system.

From the main switch individual 2300 volt power loops lead out to each light bank vault, where the current is fed into one main transformer and reduced to 110-220 volt which is fed individually to each lamp through a 32-watt 32-watt transformer. In the main vault the control panel in the control room each of the twelve circuits may be controlled through main switches in the main power vault, each bank of lights may be controlled as a unit from a main switch in each bank vault and each light may be individually controlled by a secondary switch between the main switch and each lamp. This system gives the entire installation extreme flexibility, making possible many combinations in the use of the various lights, and simplifying any necessary service or repair work.

In addition to controlling all main circuits from the primary vault, power lines for the signal deck lights are fed from this vault. There are also two secondary vaults, completely underground in front of the hangars, from which the exterior hangar lights are fed.

INSTALLATION of this equipment on the Grand Central Air Terminal has required approximately eight months, having been completed year by year to meet the demands of long winter nights and added late schedules necessitated to satisfy growing traffic.

The hangar and control lights, boundary and approach lights, and wind direction indicator lights are kept burning throughout every night and an attendant is always on duty in the control tower to switch on whatever field lights may be required by arriving or departing planes.

Lighting of the Grand Central Air Terminal is believed to be the most modern of any such installation yet made, having been made so to meet the requirements of the present air transport traffic, the handover of passengers enplaning or deplaning per month, of any air terminal in America. Approx. 6,000 passengers in September alone, 9,620 toward month. This lighting equipment is playing a big part in carrying air traffic through the terminal for the constantly growing totals. For the public interested in order to get on the ground and regularly operating from a field where such attention is given to the problem of night operations.

The Grand Central Air Terminal lighting is a big step forward in the direction which airports and airfields must go, for not only all routes and fields are sufficiently lighted to make night flying as practicable as day flying, but we expect unaccustomed public patronage of airfields. Los Angeles, with not one, but ten, lighted airports, should prove a magnet for operators who find it now practical to stop planes into this area after dark. The way in which western flying schedules are being extended regardless of the season and the percentage increase in the number of flights is an indication to the way in which the great traffic revolution which can be tapped by the provision of adequate night lighting.



HIGH POINTS in the NEWS

► **Back in town.** Aeronautical Chamber of Commerce New York air show is scheduled on May 11, with an attendance of 120,000 and sales to be expected to \$50,000 for the nine-day exposition.

► **What to see.** More than one hundred and fifty stand the South Atlantic Club. Monthly Conference at Dayton. Material specifications adopted.

► **Words come.** Receipts of May planes, wherever New York show May 11, 3, and over New England, office May 11, Army chief also received.

► **Another day on.** Introduction bill at Washington asking for design, construction, and procurement of a metal-deck aircraft for the Army.

► **Flight.** Flight design of the Curtiss Aeroplane & Motor Co., Buffalo, is to be moved to Wright Aeronautical Corp. plant, at Paterson, N. J.

► **Revised.** Operating laws on air traffic in the U.S. are to be revised and allowed under the old National Air Code.

► **No N.A.A.** now decides National Civil Aviation Council will be allowed to operate at the University of California, Germany, is awarded second Daimler-Benz award.

► **Gold medal.** Dr. Ludwig Prandtl, professor at the University of Göttingen, Germany, is awarded second Daimler-Benz award.

► **Significantly again.** By striking to 10,000 ft. or more below the 10,000 ft. level, Capt. Benji Sengulsky, chief of the 1st Air Force, has been the American record at 22,000 ft.

Airports and Airlines

► **Under aerial conditions.** This visual type radio range beacon to be used on a regular airport will be located at Bellefonte, Pa.

► **Certification.** On and after midnight of May 15, airlines must have a Certificate of Authority to operate, issued from the Department of Commerce.

► **Shower Day.** Plans Aeronautical Station Inc., to handle more flights at lot to air transport companies.

Aviation

► **Over Dials.** Forty-one steps are to be made on Indian Light Plane. From 1939 to 1941, being managed by Royal Air Force, Army and Navy of India. \$12,750 in prize money planned.

GENERAL NEWS

BRUCE F. POWELL, News Editor

Michigan Licenses 41 Schools

LANING (Mich.)—There are now 41 schools in Michigan which have been licensed by the State Board of Aeronautics, of which number are in Transport schools. Temporary licenses were recently issued to Upper Peninsula Airways, Inc., of Escanaba, and the Northern Aircraft Corp., Flint, both of which offer courses in air flying courses. Capt. Ray Collins, State Director of Aeronautics, has practically completed his inspection of airports and schools.

Curtiss Plans Changes In Factories at Buffalo

BUFFALO (N. Y.)—Plans for the development of the Curtiss Aeroplane & Motor Corp. at Buffalo, and the consolidation of all its production and assembly departments at the Kenmore plant of the company are now underway.

Plans to build another show field near the Kenmore plant, where all air plane parts including power plants are manufactured, will be moved to the Kenmore plant, where all air plane parts are now made. A new air show at a cost of \$400,000. The Kull Street property will be abandoned.

Between 1,500 and 1,700 cars, including the office staff, are now employed at the two Buffalo plants. At the Kull Street factory more than 800 cars are employed on the day shift and 300 nights, while the engine division employs between 500 and 600. The weekly payroll of the plant is between \$200,000 and \$300,000.

Error Brings Jackass Around

OAKLAND (CALIF.)—A small house jacks with wings, the wings mainly mounted on a motor stand, in the aerial now available for student pilots who cannot afford to fly in the air. The house jacks are being used by the Boeing School of Aeronautics, this city.

The first student to become the possessor of this hand-made plane of anatomy was a first who went out at night and flew without aid of his navigation lights—only to turn them on for landing purposes after he had landed.

Estimate 120,000 Saw N.Y. Air Salon

Report \$750,000 in Sales For Nine-Day Exposition

NEW YORK (N. Y.)—Having attracted a vast attendance estimated at 120,000, with first reports from exhibitors indicating that plane, glider, and engine sales during the exposition averaged approximately three quarters of a million, the affair was pronounced a success.

Plans to hold another show next year were being made by the Aeronautical Chamber of Commerce as the first New York Air Salon closed at Madison Square Garden, May 11, after running from May 1.

Air transport was the angle stressed at the Salon, and the public's response was drawn to this by the air transport machines displayed in the main arena. Specimens, delivered during the event, covered almost all travel, its development and possibilities.

Overcrowded Trip to E-11

Show visitors in most places were considerably drawn to the large planes, and one of these, the Western Air Express Sikorski E-11, closed in the main entrance, was estimated by spectators at the rate of hundred each hour.

The "most beautiful plane in the show" was the Douglas airplane, the E-32, thus making W.A.E. the originator of the latest Chevrolet Co. car in the show area.

For assembling a large host of planes to fly over New York the opening day, and for presenting air transportation by flying five times during the exposition to many hundreds of innocent leaders, the Curtiss-Wright Corp. was honored with the grand prize—this B-1 Albatross. The title of "the best aerial feature presenting air transportation during the New York air show" went with this award.

Other Events

Four planes were displayed on the exposition lawn and displayed by a Southern-style sail wing for the Curtiss-Wright Flying School, a trophy presented by the Long Island chapter of Commerce for the last airplane exhibit.

The Alumnus Company of America, which won the transportation prize at the St. Louis show, was the first to be presented by the Warren Road Motor Corp. for the most attractive display comprising more than 100 low-flying planes. The Long Island chapter of Commerce, which won the exhibition prize, was the most attractive of which won a later prize.

(Continued on Page 194)

Curtiss Marine Race Eligibility Rules Chaired

WASHINGTON (A 4) — Eligibility rules for the planes entered in the seventh annual Curtiss Marine Trophy Race, to be flown on June 13, 14, 15, May 24, have been changed somewhat from those governing last year's race. The new regulations specify that (1) planes classified as amphibians must be listed, without exception, as neither seaplane floats, (2) no Vought seaplane will be eligible, (3) planes must be stripped of mud, sand and weight flying equipment and need carry only enough fuel and oil to give a 100-mile range, (4) the classification of the race, (5) planes may be specially streamlined, provided that no permanent changes are made, and (6) standard engines, without superchargers, can only be used, as in normal operating conditions.

An aerial review of 300 planes from the Lexington, Lexington and Langley will be staged in conjunction with the Curtiss race. Flying squadrons composed of fighters placed in line will show their wits in close formation, the "High Hairs," the "Red Rippers," the "Blue Devils," the "Black Cats." There will be among the noncombat personnel. Expectations are that the "High Hairs" will go through their formation with the planes lined together with 50-ft rope lengths.

Quoted in Brother's Witness

LAST W. B. Gumbler, operations officer at Annapolis, will take the place of Al Williams (a suspended) in exhibition flying. Additional events will be a divisional exhibit (also planned) in maneuvers and tactics, para jumps from a C-47 (suspended) at New Bedford, and the flying of a Marine seaplane. Attempts are being made by the Annapolis station to get a Curtiss plane to replace the "Annapolis Sea Hawk," now scheduled June 1, 5, 10, 15, 20, 25, 30, 35, 40, 45, 50, 55, 60, 65, 70, 75, 80, 85, 90, 95, 100, 105, 110, 115, 120, 125, 130, 135, 140, 145, 150, 155, 160, 165, 170, 175, 180, 185, 190, 195, 200, 205, 210, 215, 220, 225, 230, 235, 240, 245, 250, 255, 260, 265, 270, 275, 280, 285, 290, 295, 300, 305, 310, 315, 320, 325, 330, 335, 340, 345, 350, 355, 360, 365, 370, 375, 380, 385, 390, 395, 400, 405, 410, 415, 420, 425, 430, 435, 440, 445, 450, 455, 460, 465, 470, 475, 480, 485, 490, 495, 500, 505, 510, 515, 520, 525, 530, 535, 540, 545, 550, 555, 560, 565, 570, 575, 580, 585, 590, 595, 600, 605, 610, 615, 620, 625, 630, 635, 640, 645, 650, 655, 660, 665, 670, 675, 680, 685, 690, 695, 700, 705, 710, 715, 720, 725, 730, 735, 740, 745, 750, 755, 760, 765, 770, 775, 780, 785, 790, 795, 800, 805, 810, 815, 820, 825, 830, 835, 840, 845, 850, 855, 860, 865, 870, 875, 880, 885, 890, 895, 900, 905, 910, 915, 920, 925, 930, 935, 940, 945, 950, 955, 960, 965, 970, 975, 980, 985, 990, 995, 1000, 1005, 1010, 1015, 1020, 1025, 1030, 1035, 1040, 1045, 1050, 1055, 1060, 1065, 1070, 1075, 1080, 1085, 1090, 1095, 1100, 1105, 1110, 1115, 1120, 1125, 1130, 1135, 1140, 1145, 1150, 1155, 1160, 1165, 1170, 1175, 1180, 1185, 1190, 1195, 1200, 1205, 1210, 1215, 1220, 1225, 1230, 1235, 1240, 1245, 1250, 1255, 1260, 1265, 1270, 1275, 1280, 1285, 1290, 1295, 1300, 1305, 1310, 1315, 1320, 1325, 1330, 1335, 1340, 1345, 1350, 1355, 1360, 1365, 1370, 1375, 1380, 1385, 1390, 1395, 1400, 1405, 1410, 1415, 1420, 1425, 1430, 1435, 1440, 1445, 1450, 1455, 1460, 1465, 1470, 1475, 1480, 1485, 1490, 1495, 1500, 1505, 1510, 1515, 1520, 1525, 1530, 1535, 1540, 1545, 1550, 1555, 1560, 1565, 1570, 1575, 1580, 1585, 1590, 1595, 1600, 1605, 1610, 1615, 1620, 1625, 1630, 1635, 1640, 1645, 1650, 1655, 1660, 1665, 1670, 1675, 1680, 1685, 1690, 1695, 1700, 1705, 1710, 1715, 1720, 1725, 1730, 1735, 1740, 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8385, 8390, 8395, 8400, 8405, 8410, 8415, 8420, 8425, 8430, 8435, 8440, 8445, 8450, 8455, 8460, 8465, 8470, 8475, 8480, 8485, 8490, 8495, 8500, 8505, 8510, 8515, 8520, 8525, 8530, 8535, 8540, 8545, 8550, 8555, 8560, 8565, 8570, 8575, 8580, 8585, 8590, 8595, 8600, 8605, 8610, 8615, 8620, 8625, 8630, 8635, 8640, 8645, 8650, 8655, 8660, 8665, 8670, 8675, 8680, 8685, 8690, 8695, 8700, 8705, 8710, 8715, 8720, 8725, 8730, 8735, 8740, 8745, 8750, 8755, 8760, 8765, 8770, 8775, 8780, 8785, 8790, 8795, 8800, 8805, 8810, 8815, 8820, 8825, 8830, 8835, 8840, 8845, 8850, 8855, 8860, 8865, 8870, 8875, 8880, 8885, 8890, 8895, 8900, 8905, 8910, 8915, 8920, 8925, 8930, 8935, 8940, 8945, 8950, 8955, 8960, 8965, 8970, 8975, 8980, 8985, 8990, 8995, 9000, 9005, 9010, 9015, 9020, 9025, 9030, 9035, 9040, 9045, 9050, 9055, 9060, 9065, 9070, 9075, 9080, 9085, 9090, 9095, 9100, 9105, 9110, 9115, 9120, 9125, 9130, 9135, 9140, 9145, 9150, 9155, 9160, 9165, 9170, 9175, 9180, 9185, 9190, 9195, 9200, 9205, 9210, 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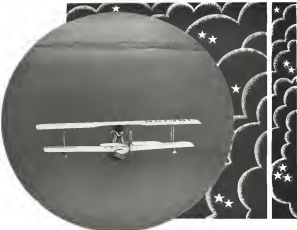
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Generalized Adversities

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